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Instability of co-flow in a Hele-Shaw cell with cross-flow varying thickness

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(Received xx; revised xx; accepted xx)

We analyse the stability of the interface between two immiscible fluids both flowing in the horizontal direction in a thin cell with vertically varying gap width. The dispersion relation for the growth rate of each mode is derived. The stability is approximately determined by the sign of a simple expression, which incorporates the density difference between the fluids and the effect of surface tension in the along- and cross-cell directions. The latter arises from the varying channel width: if the non-wetting fluid is in the thinner part of the channel, the interface is unstable as it will preferentially migrate into the thicker part. The density difference may suppress or complement this effect. The system is always stable to sufficiently large wavenumbers owing to the along-flow component of surface tension.

1. Introduction

The parallel co-flow of two fluids occurs in many industrial, biological and environmental processes. It is often important to understand the interfacial instability and develop strategies to control it. Frequently, these flows occur in thin channels whose thickness varies in the cross-flow direction. Examples include: the flow of cement and drilling fluid within a casing pipe of a subsurface well, where the intermingling of cement and mud can produce poorly sealed wells with the attendant risks of leakage; the flow of coatings in the corner region along the line of intersection between two planes, where the displacement of air limits the formation of non-coated zones (Weislogel & Lichter 1996); flows of reactants in microfluidic channels (Sauer 1987; Huang *et al.* 2018); and the displacement of water by CO₂ in permeable channels used for CO₂ sequestration, where intermingling may enhance the efficiency of the sequestration (Woods & Mingotti 2016). For the last example of CO₂ sequestration, the pore-scale dynamics, which are controlled by capillary effects and the interpore geometry are not fully understood but can have a significant effect on macroscale mechanisms such as the flux and the residual trapping of the CO₂, which are key to estimating storage efficiency (Zhao *et al.* 2019; Benham *et al.* 2021). If the initial flow involves the displacement of one fluid by a second along the channel, the displacing fluid will migrate along the wide part of the channel, stretching out the interface; at long times, the interface is approximately directed along the channel, and the flow evolves to the co-flowing geometry of the present problem (c.f. Woods & Mingotti 2016; Mortimer & Woods 2021).

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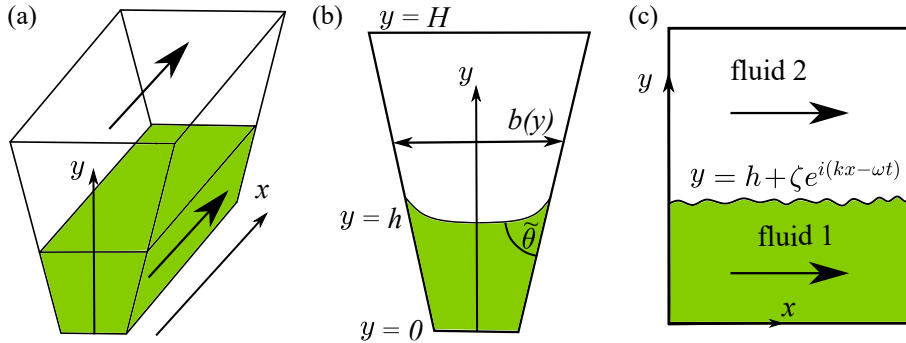


Figure 1: (a) Schematic of the setup. Large arrows indicate the flow direction. (b) Cross-section perpendicular to the x direction. (c) Cross-section in the x direction.

36 We investigate how the stability of the interface between the two fluids in a thin cell with
 37 vertically varying gap width is controlled by cross-layer buoyancy and capillary effects (figure
 38 1). We assume that inertia plays a negligible role in the base flow. It has been shown that when
 39 inertia plays a significant role and the two fluids in the Hele-Shaw cell have significantly
 40 different viscosities, the shear-controlled Kelvin-Helmholtz instability may occur as has
 41 been observed experimentally (Zeybek & Yortsos 1992; Gondret & Rabaud 1997; Rabaud
 42 & Moisy 2020). In wider channels, it has been shown that even at zero Reynolds number the
 43 vertical shear associated with the no-slip boundaries at the top and bottom can give rise to
 44 interfacial instabilities in the co-flow of two fluids of different viscosity (Yih 1967). Similar
 45 behaviour can occur in two-layer gravity-driven flow (Loewenherz & Lawrence 1989).

46 In the present work, we consider a laterally thin cell in which the vertically varying velocity
 47 arises from variations in the cell width. The stability is primarily controlled by the density
 48 difference between the fluids and surface tension. It is well-established that the along-flow
 49 component of surface tension stabilises larger wavenumbers. The combination of surface
 50 tension and the cross-cell variation in thickness introduces a new (de)stabilising process
 51 for small wavenumbers in the case that the (non-)wetting fluid is in the thinner part of the
 52 channel. This effect may complement or suppress the effect of a density difference between
 53 the two fluids on the stability of the interface.

54 The impact of variations in the surface tension associated with variations in the channel
 55 width have been explored in detail for the related problem in which an input fluid displaces
 56 an ambient fluid in a cell whose width varies in the direction of flow (Homsy 1987; Al-
 57 Housseiny *et al.* 2012; Dias & Miranda 2013; Grenfell-Shaw & Woods 2017). These studies
 58 have identified that the effect of cross-cell curvature can complement or suppress the classical
 59 Saffman-Taylor instability (Saffman & Taylor 1958).

60 2. Formulation

61 The flow and the cell geometry is illustrated in figure 1. The cell occupies $0 < y < H$ and
 62 has width, $b(y)$, that varies in the vertical direction,

$$63 \quad b(y) = b_0 + \alpha y, \quad (2.1)$$

64 where α represents the inclination of the cell walls, which may be positive or negative, and
 65 $\alpha > -b_0/H$ so that the cell width is non-negative. Flow is driven in both fluids in the x
 66 direction by a background pressure gradient with magnitude G . For relatively slow flows, we
 67 can apply the lubrication approximation, which is to say that the leading order velocities are
 68 independent of x . Under this assumption, the momentum and continuity equations for the

69 gap-averaged velocity in each fluid take the form (equation 11 of Gondret & Rabaud 1997)

$$70 \quad \frac{\partial \bar{\mathbf{u}}}{\partial t} + \bar{\mathbf{u}} \cdot \nabla \bar{\mathbf{u}} = -\frac{1}{\rho} \nabla p - \frac{12\mu}{\rho b^2} \bar{\mathbf{u}} - g \mathbf{e}_y, \quad \nabla \cdot (b\bar{\mathbf{u}}) = 0, \quad (2.2)$$

71 where, $\nabla = (\partial/\partial x, \partial/\partial y)$, \mathbf{e}_y is the unit vector in the y direction and μ and ρ are the fluid
72 viscosity and density, respectively. Also, p is the pressure, g represents gravity, which acts
73 in the negative y direction and $\bar{\mathbf{u}} = (\bar{u}, \bar{v})$ is the width-averaged velocity in the x and y
74 directions. The boundary conditions are no-flux at the top and bottom of the channel: $\bar{v} = 0$,
75 whilst at the fluid-fluid interface, $y = y_I$, the velocity in each fluid satisfies the kinematic
76 boundary condition

$$77 \quad \frac{\partial y_I}{\partial t} + \bar{u} \frac{\partial y_I}{\partial x} = \bar{v}. \quad (2.3)$$

78 In addition, there is a pressure jump at the interface associated with its curvature, $\kappa = \nabla^2 y_I$,
79 given by

$$80 \quad \Delta p = \gamma \kappa, \quad (2.4)$$

81 where γ represents surface tension. The unperturbed steady base flow is given by

$$82 \quad \bar{u} = U_0(y) = \frac{b(y)^2 G}{12\mu}, \quad \bar{v} = 0, \quad p = P_0(x, y) = -\rho g y - Gx + \text{const}. \quad (2.5)$$

83 The fluids are immiscible and the location of the fluid-fluid interface, $y_I = h$ is a constant for
84 the case of steady flow, which depends on the channel angle, the relative flux in each layer
85 and the viscosity ratio. Although the curvature of the interface in the along-flow direction
86 vanishes since y_I is independent of x , there is curvature in the cross-channel direction owing
87 to the contact angle at the wall and the varying channel width (figure 1b). Hence, there is a
88 pressure jump at the interface, which is independent of x , and the constants in P_0 in the base
89 flow are different in the two fluids.

90 We consider perturbations to the interface and steady base flow of the form

$$91 \quad y_I = h + \zeta e^{i(kx - \omega t)} \quad (2.6)$$

$$92 \quad (\bar{u}, \bar{v}) = (U_0(y), 0) + (u(y), v(y)) e^{i(kx - \omega t)}, \quad (2.7)$$

$$93 \quad p = P_0(x, y) + p(y) e^{i(kx - \omega t)}. \quad (2.8)$$

95 where ζ , $u(y)$, $v(y)$ and $p(y)$ are assumed to be small. We seek to determine the stability of
96 such perturbations. The linearised governing equations in each fluid are

$$97 \quad -i\omega u + ikU_0 u + vU_0' = \frac{-ikp}{\rho} - \frac{12\mu u}{\rho b^2} \quad (2.9)$$

$$98 \quad -i\omega v + ikU_0 v = -\frac{p'}{\rho} - \frac{12\mu v}{\rho b^2} \quad (2.10)$$

$$100 \quad ikub + (vb)' = 0, \quad (2.11)$$

101 where a prime (') denotes differentiation with respect to y . We eliminate p' to obtain

$$102 \quad -i\omega u' + ik(U_0 u)' + (vU_0')' - k\omega v + k^2 U_0 v = \frac{12ik\mu v}{\rho b^2} - \frac{12\mu(u/b^2)'}{\rho}. \quad (2.12)$$

103 Also, continuity yields

$$104 \quad u = \frac{i(vb)'}{kb} = \frac{i}{k} \left(v' + \frac{v\alpha}{b} \right), \quad (2.13)$$

105 which can be used to eliminate $u(y)$ from equation (2.12) and obtain a differential equation
106 for $v(y)$,

$$107 \quad A(y)v'' + B(y)v' + C(y)v = 0, \quad (2.14)$$

108 where the coefficients are defined below for the dimensionless analogue.

109 2.1. Non-dimensionalisation

110 To scale the system, we use the tank height, H as the length-scale and the time-scale is
111 $T = \mu_2/(GH)$. The pressure scale is GH . We write

$$112 \quad (\hat{x}, \hat{y}) = (x, y)/H, \quad \hat{k} = Hk, \quad \hat{w} = Tw, \quad \hat{b}(\hat{y}) = b(y)/H = \hat{b}_0 + \alpha\hat{y}, \quad (2.15)$$

113 with $\hat{b}_0 = b_0/H$, $\hat{h} = h/H$ and $\hat{\zeta} = \zeta/H$. The upper fluid is labelled fluid 2, whilst the
114 lower is labelled fluid 1 (figure 1). Henceforth, all quantities are dimensionless unless stated
115 otherwise and we discard the hat notation. The dimensionless equation in fluid $j = 1, 2$
116 becomes

$$117 \quad A^{(j)}(y)v_{yy}^{(j)} + B^{(j)}(y)v_y^{(j)} + C^{(j)}(y)v^{(j)} = 0, \quad (2.16)$$

118 with coefficients,

$$119 \quad A^{(j)} = \frac{\omega b(y)^4}{k} - \frac{M_j b(y)^6}{12} + \frac{ib(y)^2 R_j \mathcal{A}}{k} \quad (2.17)$$

$$120 \quad B^{(j)} = \frac{\omega \alpha b(y)^3}{k} - \frac{M_j \alpha b(y)^5}{12} - \frac{i \alpha b(y) R_j \mathcal{A}}{k} \quad (2.18)$$

$$121 \quad C^{(j)} = \frac{-\omega \alpha^2 b(y)^2}{k} + \frac{M_j \alpha^2 b(y)^4}{12} - k \omega b(y)^4 + \frac{M_j k^2 b(y)^6}{12} - ikb(y)^2 R_j \mathcal{A} - \frac{3i \alpha^2 R_j \mathcal{A}}{k}, \quad (2.19)$$

123 where $M_1 = M$, $M_2 = 1$ and $R_1 = R$, $R_2 = M$, and we have introduced the following
124 dimensionless parameters,

$$125 \quad M = \frac{\mu_2}{\mu_1}, \quad R = \frac{\rho_2}{\rho_1}, \quad \mathcal{A} = \frac{12\mu_1\mu_2}{\rho_2 H^3 G}, \quad (2.20)$$

126 which respectively represent the viscosity ratio, the density ratio and the importance of
127 viscous drag relative to inertia; \mathcal{A} is inversely proportional to a Reynolds number.

128 2.2. Boundary conditions

129 The perturbed no-flux boundary conditions are

$$130 \quad v^{(1)}(0) = v^{(2)}(1) = 0. \quad (2.21)$$

131 The kinematic boundary conditions (2.3) at the fluid-fluid interface become

$$132 \quad v^{(1)}(h) = i\zeta \left[\frac{b(h)^2 M}{12} k - \omega \right], \quad v^{(2)}(h) = i\zeta \left[\frac{b(h)^2}{12} k - \omega \right]. \quad (2.22)$$

133 The dynamic boundary condition at the interface accounts for a jump in pressure associated
134 with surface tension and curvature. This comprises two contributions: the along-channel
135 curvature and the cross-channel curvature. The former is proportional to the second derivative
136 of the interface y_I in the x direction, which furnishes a term proportional to k^2 . Treating
137 the cross-channel curvature requires more care. The contact angle, $\hat{\theta}$, is defined as the angle
138 between fluid 1 and the channel wall (figure 1b). In a cell with inclined walls, the radius
139 of curvature is adjusted from the case of a parallel sided cell and we define an effective

140 contact angle, $\theta = \tilde{\theta} - \phi$, where $\tan \phi = \alpha/2$ (Park & Homsy 1984; Romero & Yost 1996;
 141 Grenfell-Shaw & Woods 2017). The discontinuity in the perturbed pressures at the interface
 142 is then given by

$$143 \quad p_2(h) - p_1(h) = C\zeta \left[R - 1 - Bo^{-1} \left(k^2 + \frac{2\alpha \cos \theta}{b(h)^2} \right) \right], \quad (2.23)$$

144 where we have introduced the following dimensionless groups

$$145 \quad C = \frac{\rho_1 g}{G}, \quad Bo = \frac{g \rho_1 H^2}{\gamma}. \quad (2.24)$$

146 We use the dimensionless analogues of equations (2.9) and (2.13) to obtain the dimensionless
 147 pressures in terms of the vertical velocity, v , in each fluid,

$$148 \quad p_j = \frac{12i\omega}{k^2 R_j M_j \mathcal{A}} \frac{(v^{(j)} b)'}{b} - \frac{i}{k} \frac{b(v^{(j)} b)'}{\mathcal{A} R_j} + \frac{2i\alpha b v^{(j)}}{k \mathcal{A} R_j} - \frac{12}{M_j k^2} \frac{(v^{(j)} b)'}{b^3}. \quad (2.25)$$

149 We substitute the pressures into the dynamic boundary condition (2.23) to obtain

$$150 \quad \alpha v^{(1)} \left(\frac{\omega}{kb} + \frac{Mb}{12} + \frac{iR\mathcal{A}}{kb^3} \right) + \frac{dv^{(1)}}{dy} \left(\frac{\omega}{k} - \frac{Mb^2}{12} + \frac{iR\mathcal{A}}{kb^2} \right) \\
 151 \quad - R\alpha v^{(2)} \left(\frac{\omega}{kb} + \frac{b}{12} + \frac{iM\mathcal{A}}{kb^3} \right) - R \frac{dv^{(2)}}{dy} \left(\frac{\omega}{k} - \frac{b^2}{12} + \frac{iM\mathcal{A}}{kb^2} \right) \\
 152 \quad = \frac{kRM\mathcal{A}i}{12} C\zeta \left[R - 1 - Bo^{-1} \left(k^2 + \frac{2\alpha \cos \theta}{b^2} \right) \right], \quad (2.26)$$

154 where the suppressed argument of $v^{(1)}$, $v^{(2)}$, $dv^{(1)}/dy$, $dv^{(2)}/dy$ and b is $y = h$.

155 3. Solution method

156 The system for $v(y)$ comprises two second order linear ODEs in $0 < y < h$ and $h < y < H$
 157 (2.16) and four boundary conditions: no flux at the top and bottom boundaries (2.21) and
 158 the kinematic and dynamic boundary conditions (2.22, 2.26) at the interface. We note that
 159 the two equations for the kinematic condition (2.22) can be used to eliminate ζ from the
 160 problem. To solve this system, we first simplify the problem by writing $\bar{b}(y) = kb(y)/\alpha$ and
 161 obtain the following equation for $v^{(j)}(\bar{b})$,

$$162 \quad A^{(j)}(y)v_{\bar{b}\bar{b}}^{(j)} + B^{(j)}(y)v_{\bar{b}}^{(j)} + C^{(j)}(y)v^{(j)} = 0, \quad (3.1)$$

163 with

$$164 \quad A^{(j)} = a_6^{(j)} \bar{b}^6 + a_4^{(j)} \bar{b}^4 + \bar{b}^2 \quad (3.2)$$

$$165 \quad B^{(j)} = a_6^{(j)} \bar{b}^5 + a_4^{(j)} \bar{b}^3 - \bar{b} \quad (3.3)$$

$$166 \quad C^{(j)} = -a_6^{(j)} (\bar{b}^6 + \bar{b}^4) - a_4^{(j)} (\bar{b}^4 + \bar{b}^2) - \bar{b}^2 - 3, \quad (3.4)$$

168 where we have introduced the following for each fluid,

$$169 \quad a_6^{(j)} = \frac{-M_j \alpha^4}{12iR_j \mathcal{A} k^3}, \quad a_4^{(j)} = \frac{\omega \alpha^2}{iR_j \mathcal{A} k^2}. \quad (3.5)$$

170 The general solution for v in each fluid is given by a linear combination of two independent

171 power series, $\Phi^{(j)}(\bar{b})$ and $\Psi^{(j)}(\bar{b})$, whose coefficients are determined by Frobenius' method
 172 (given in Appendix A). The velocities are ($j = 1, 2$)

$$173 \quad v^{(j)}(y) = c^{(j)} \left(\frac{\Phi^{(j)}(\bar{b}(y))}{\Phi^{(j)}(\bar{b}(j-1))} - \frac{\Psi^{(j)}(\bar{b}(y))}{\Psi^{(j)}(\bar{b}(j-1))} \right), \quad (3.6)$$

174 where $c^{(j)}$ are constants and we have used the no-flux boundary conditions at the base and
 175 the top ($y = j - 1$). The dynamic boundary condition may be written as (at $y = h$),

$$176 \quad \frac{(v^{(1)}\bar{b})_{\bar{b}}}{\bar{b}^3} + a_4^{(1)} \frac{(v^{(1)}\bar{b})_{\bar{b}}}{\bar{b}} + a_6^{(1)} \left(\frac{v^{(1)}}{\bar{b}} \right)_{\bar{b}} \bar{b}^3 \quad (3.7)$$

$$177 \quad - M \left[\frac{(v^{(2)}\bar{b})_{\bar{b}}}{\bar{b}^3} + a_4^{(2)} \frac{(v^{(2)}\bar{b})_{\bar{b}}}{\bar{b}} + a_6^{(2)} \left(\frac{v^{(2)}}{\bar{b}} \right)_{\bar{b}} \bar{b}^3 \right] \quad (3.8)$$

$$178 \quad = \frac{iM\alpha^2 v^{(1)}}{12k\omega} C \left[R - 1 - Bo^{-1} \left(k^2 + \frac{2\alpha \cos \theta}{b^2} \right) \right] \left(\frac{a_6^{(1)}}{a_4^{(1)}} \bar{b}^2 + 1 \right)^{-1}. \quad (3.9)$$

180 The kinematic boundary condition may be written as

$$181 \quad v^{(1)} \left(\frac{a_6^{(2)}}{a_4^{(2)}} \bar{b}^2 + 1 \right) = v^{(2)} \left(\frac{a_6^{(1)}}{a_4^{(1)}} \bar{b}^2 + 1 \right). \quad (3.10)$$

182 Combining the velocities with these two boundary conditions furnishes the following
 183 dispersion relation

$$184 \quad D^{(1)}E^{(1)} - ME^{(2)}D^{(2)} - S = 0, \quad (3.11)$$

185 where

$$186 \quad E^{(j)} = \frac{t_j}{\bar{b}^2} + \frac{1}{\bar{b}^3} + a_4^{(j)} \left(t_j + \frac{1}{\bar{b}} \right) + a_6^{(j)} \left(t_j \bar{b}^2 - \bar{b} \right), \quad (3.12)$$

$$187 \quad D^{(j)} = \frac{a_6^{(j)}}{a_4^{(j)}} \bar{b}^2 + 1, \quad (3.13)$$

$$188 \quad S = \frac{iM\alpha^2}{12k\omega} C \left[R - 1 - Bo^{-1} \left(k^2 + 2\alpha \cos \theta / b(h)^2 \right) \right], \quad (3.14)$$

$$189 \quad t_1 = \frac{\Phi_{\bar{b}}^{(1)}(\bar{b}(h))/\Phi^{(1)}(\bar{b}(0)) - \Psi_{\bar{b}}^{(1)}(\bar{b}(h))/\Psi^{(1)}(\bar{b}(0))}{\Phi^{(1)}(\bar{b}(h))/\Phi^{(1)}(\bar{b}(0)) - \Psi^{(1)}(\bar{b}(h))/\Psi^{(1)}(\bar{b}(0))}, \quad (3.15)$$

$$190 \quad t_2 = \frac{\Phi_{\bar{b}}^{(2)}(\bar{b}(h))/\Phi^{(2)}(\bar{b}(1)) - \Psi_{\bar{b}}^{(2)}(\bar{b}(h))/\Psi^{(2)}(\bar{b}(1))}{\Phi^{(2)}(\bar{b}(h))/\Phi^{(2)}(\bar{b}(1)) - \Psi^{(2)}(\bar{b}(h))/\Psi^{(2)}(\bar{b}(1))}, \quad (3.16)$$

192 where i is the imaginary unit. For any value of k and the dimensionless parameters, we may
 193 obtain a solution for the growth rate, ω that satisfies equation (3.11) (e.g. figure 2a).

194 4. Analysis

195 The terms in the square brackets in S (equation 3.14) correspond to the pressure jump at the
 196 interface and we define

$$197 \quad J = R - 1 - Bo^{-1} \left[k^2 + 2\alpha \cos \theta / b(h)^2 \right]. \quad (4.1)$$

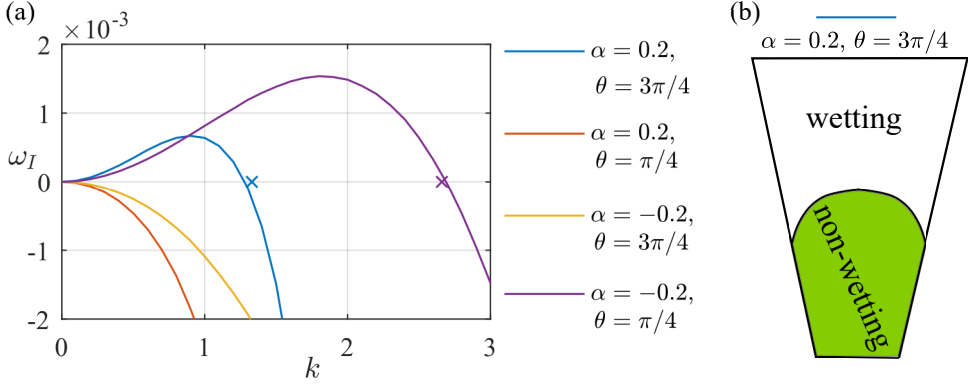


Figure 2: (a) Growth rate, ω_I as a function of wave number k in the case of equal density fluids ($R = 1$). The curves are calculated using the method in §3. The critical wavenumbers predicted by (4.3) are shown as crosses. We use $Bo = 5$, $h = 0.5$, $b_0 = 0.3$, $M = 2$, $\mathcal{A} = 1$ and $C = 1$. (b) Schematic corresponding to the blue curve.

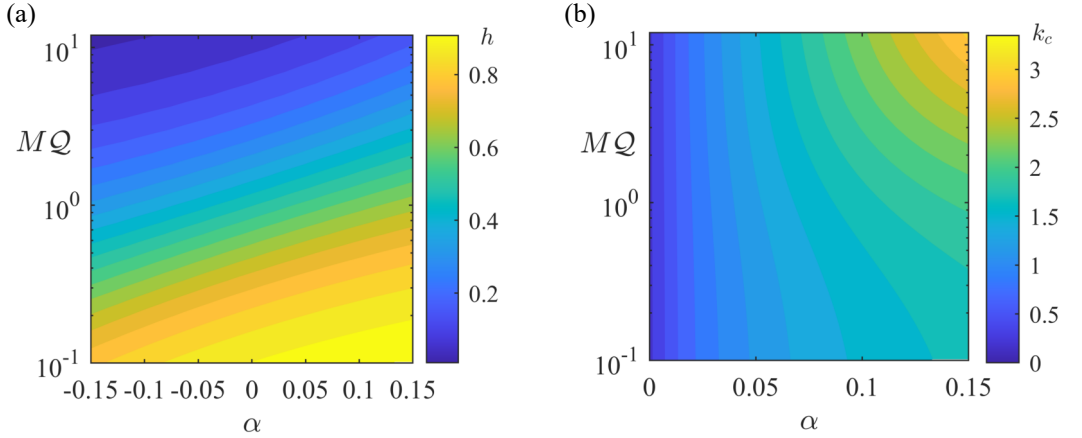


Figure 3: (a) The interface height, h , as a function of the relative flux and viscosity, and the channel angle, α for a fixed dimensionless channel area of 0.2. (b) The corresponding critical wavenumber, k_c above which the system is stable according to (4.3) for $\theta = 3\pi/4$. Note that the system is stable for all k for $\alpha \leq 0$.

198 In general, $J > 0$ is associated with instability and $J < 0$ is associated with stability. The first
 199 term, $R - 1$ represents the density difference between the fluids. It stabilizes the interface for
 200 $R < 1$ and destabilizes it for $R > 1$. The term, $-Bo^{-1}k^2$ stabilizes the interface; it arises from
 201 surface tension suppressing the curvature in the along-channel (x) direction. The final term
 202 $-2Bo^{-1}\alpha \cos \theta/b(h)^2$ is associated with surface tension acting on the curvature across the
 203 thickness of the cell. It drives or suppresses an instability depending on whether the wetting
 204 or non-wetting fluid is in the thinner part of the channel. This corresponds to the sign of
 205 α and the sign of $\cos \theta$. To interpret the instability, we consider the simpler cases of equal
 206 density in §4.1, parallel walls in §4.2 before returning to the full problem in §4.3.

207 4.1. Equal density ($R = 1$)

208 In the case of equal density fluids, $R = 1$, the pressure jump reduces to

$$209 J = -Bo^{-1} [k^2 + 2\alpha \cos \theta / b(h)^2]. \quad (4.2)$$

210 For small wavenumber, k , the cross-cell surface tension term controls the stability as
 211 demonstrated in figure 2a. The red and yellow curves correspond to the non-wetting fluid
 212 occupying the thicker part of the channel and the system is stable as this fluid will remain in
 213 the thicker part of the channel. The blue and purple curves represent the converse situation in
 214 which the non-wetting fluid is in the thinner part of the channel and it will move to the wider
 215 side of the channel leading to an instability (see figure 2b). For larger wavenumbers, the
 216 along-channel term stabilizes the interface. The along-channel and cross-channel curvature
 217 terms balance ($Bo^{-1}k^2 + 2Bo^{-1}\alpha \cos \theta / b(h)^2 = 0$) at a critical wavenumber,

$$218 \quad k_c = \frac{\sqrt{-2\alpha \cos \theta}}{b(h)}. \quad (4.3)$$

219 For $k > k_c$, we anticipate that the system is stable. The critical wave numbers are shown
 220 by crosses for the two unstable setups in figure 2, demonstrating good agreement with
 221 the predictions from §3. The small discrepancy between the prediction of J (4.3) and the
 222 numerical results arises because of the physical effects, such as inertia, that are incorporated
 223 in the numerics but are neglected when using J as an approximation of the stability criterion.

224 In many settings, it is important to understand how the instability depends on the flux in
 225 each layer. To analyse this, we calculate the relative flux, Q of the top to the bottom layer,

$$226 \quad Q = \frac{Q_2}{Q_1} = M^{-1} \frac{(b_0 + \alpha)^3 - (b_0 + \alpha h)^3}{(b_0 + \alpha h)^3 - b_0^3}. \quad (4.4)$$

227 The quantity $MQ = \mu_2 Q_2 / (\mu_1 Q_1)$ depends only on α , h and b_0 . We consider channels of
 228 fixed dimensionless area so that

$$229 \quad \int_0^1 b(y) dy = b_0 + \alpha/2 = \text{constant}. \quad (4.5)$$

230 For a given channel area and relative flux, Q , we can calculate the interface height h as a
 231 function of the channel angle α (see figure 3a for the case with dimensionless area 0.2).
 232 We can also calculate the critical wavenumber, k_c by using (4.3) (see figure 3b for the case
 233 $\theta = 3\pi/4$).

234 4.2. Parallel cell walls ($\alpha = 0$)

235 In the case that the cell walls are parallel,

$$236 \quad J = R - 1 - Bo^{-1}k^2. \quad (4.6)$$

237 For small k , the stability is controlled solely by the density ratio, R . In the case that $R < 1$, the
 238 system is stabilised by the density difference and there are no destabilizing effects (Gondret
 239 & Rabaud 1997). For $R > 1$, the Rayleigh-Taylor instability is stabilised for large k owing to
 240 the along cell surface tension. Neutral stability is given by

$$241 \quad k_c = \sqrt{Bo(R - 1)}. \quad (4.7)$$

242 Figure 4 shows the growth rate as a function of wavenumber (obtained in §3). For $R > 1$, the
 243 critical wavenumber prediction is indicated by crosses, showing good agreement.

244 4.3. Competition between density difference and surface tension

245 The cross-cell surface tension effect may be nullified or complemented by buoyancy,
 246 depending on whether the wetting or non-wetting fluid is denser. The critical value of R

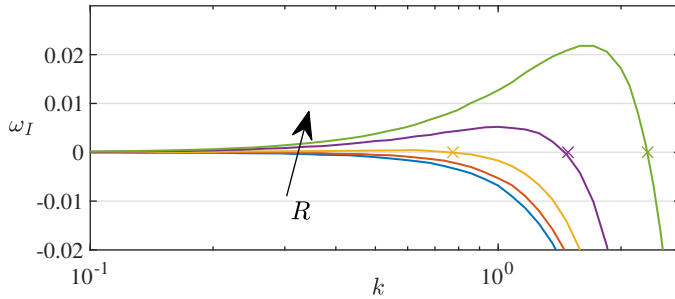


Figure 4: Growth rate, ω_I as a function of wave number, k , in the case of parallel cell walls ($\alpha = 0$) calculated using the method of §3. The density ratio is $R = 0.4, 0.8, 1.6, 3.2, 6.4$. The crosses correspond to the critical wavenumber for neutral stability for $R > 1$ given by (4.7). We use $h = 0.5, b_0 = 0.3, M = 2, \mathcal{A} = 1, C = 1$ and $Bo = 1$.

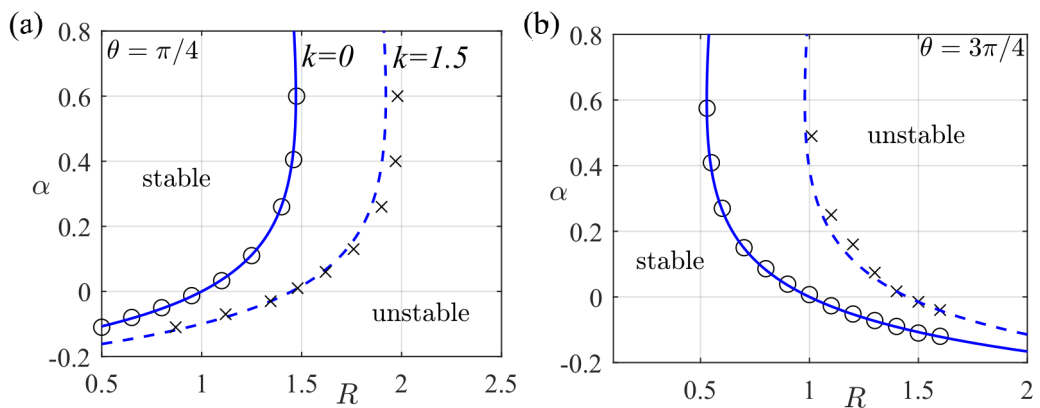


Figure 5: Neutral stability curves for (a) $\theta = \pi/4$ and (b) $\theta = 3\pi/4$. Blue lines show the predictions of (4.8) for $k = 0$ and $k = 1.5$. Circles and crosses show the results from §3 for $k = 0.2, 1.5$, respectively. We use $Bo = 5, h = 0.5, b_0 = 0.3, M = 2, \mathcal{A} = 1$ and $C = 1$.

247 corresponding to neutral stability is (see 4.1)

$$248 \quad R_c(k) = 1 + \frac{k^2}{Bo} + \frac{2\alpha \cos \theta}{Bo(b_0 + \alpha h)^2}. \quad (4.8)$$

249 The system is stable for all wavenumbers when $R < R_c(0)$, which is shown as a continuous
 250 blue line in figure 5. The results from §3 for neutral stability for $k = 0.2$ are plotted as black
 251 circles showing good agreement. A comparison is also shown for $k = 1.5$ as a broken blue
 252 line.

253 Figure 6 shows the critical density ratio $R_c(0)$ as a function of the relative flux, and the
 254 channel inclination α in the case of a constant cell area, 0.2. The interface height, h is
 255 obtained from (4.4). When α is small, the critical density ratio becomes independent of the
 256 relative flux (and hence the interface height, h) because wetting effects become unimportant.

257 5. Conclusion

258 We have obtained the dispersion relation for the co-flow of two immiscible fluids in a Hele-
 259 Shaw cell with vertically varying gap width. The stability of the system is accurately predicted

10

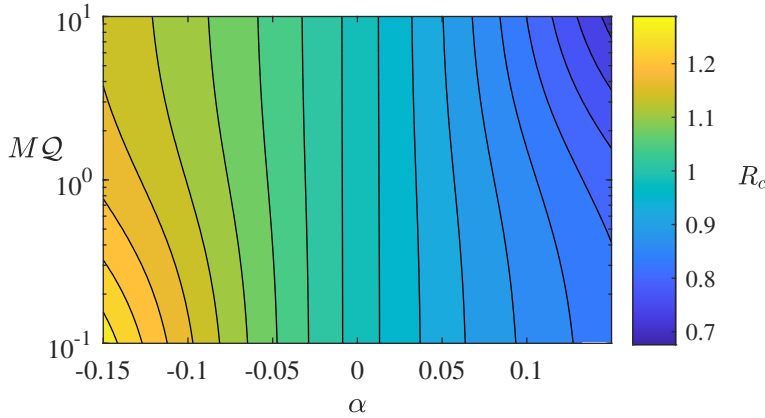


Figure 6: Critical value of R corresponding to neutral stability from (4.8) as a function of cell wall inclination, α , and relative flux QM for $k = 0$, $\theta = 3\pi/4$, $Bo = 5$. The cell area is fixed as 0.2 and the interface position is given in figure 3a.

260 by the sign of the quantity

$$261 \quad J = R - 1 - Bo^{-1} [k^2 + 2\alpha \cos \theta / b(h)^2]. \quad (5.1)$$

262 The last term, associated with channel wall inclination, represents the preference of the
 263 non-wetting fluid to occupy the thicker part of the channel. The interface is stable when the
 264 fluids have equal density and the wetting fluid occupies the thinner part of the channel. A
 265 density difference may complement or oppose this effect. We have obtained critical values of
 266 the density ratio, R , below which the system is stable to all wave-numbers. Our results also
 267 provide a basis for exploring the stability of important but more complex situations such as
 268 cells with elastic walls and cells whose vertical structure varies in the horizontal direction
 269 (e.g. Pihler-Puzović *et al.* 2013).

270 Declaration of Interests: The authors report no conflict of interest.

271 Appendix A. Coefficients for Frobenius' method

272 In either fluid, the governing equation takes the form

$$273 \quad (\bar{b}^2 + a_4 \bar{b}^4 + a_6 \bar{b}^6) v_{\bar{b}\bar{b}} + (-\bar{b} + a_4 \bar{b}^3 + a_6 \bar{b}^5) v_{\bar{b}} + (-3 + (-1 - a_4) \bar{b}^2 - (a_4 + a_6) \bar{b}^4 - a_6 \bar{b}^6) v = 0. \quad (A 1)$$

274 The indicial polynomial is $n^2 - 2n - 3 = 0$, which has solutions, $n = 3$ and $n = -1$. We write
 275 the first power series of $v(\bar{b})$ as

$$276 \quad \Phi(x) = x^3 \sum_0^{\infty} P_n x^n, \quad (A 2)$$

277 with $P_0 = 1$ and the recurrence relation

$$278 \quad P_n [n^2 + 4n] + P_{n-2} [n(n+2)a_4 - 1] + P_{n-4} [a_6(n^2 - 2n) - a_4] - P_{n-6} a_6 = 0 \quad (A 3)$$

279 The second independent power series solution is given by

$$280 \quad \Psi(x) = \Phi(x) \log x + x^{-1} \sum_0^{\infty} Q_n x^n, \quad (A 4)$$

281 where $Q_0 = 16/(4a_4 - 1)$, $Q_2 = -Q_0/4$, $Q_4 = 1$ and

$$282 \quad 0 = (n^2 - 4n)Q_n + Q_{n-2}[a_4(n-2)(n-4) - 1] + Q_{n-4}[a_6(n-4)(n-6) - a_4] \quad (\text{A } 5)$$

$$283 \quad - Q_{n-6}a_6 + P_{n-4}(2n-4) + P_{n-6}a_4(2n-6) + P_{n-8}a_6(2n-10). \quad (\text{A } 6)$$

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